A PERSPECTIVE OF NIOBIUM IN SUPERALLOYS

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Introduction

The central objective of this paper is to characterize the metallurgical behavior of niobium as an alloying element in austenitic superalloys. To accomplish this, we will first define, then describe superalloys in some detail as background. Next, niobium itself will be reviewed with respect to its properties appropriate for superalloy constitution. Then the specific function of niobium as additive to nickel, cobalt and iron-based superalloys will be established and its broad effects summarized.

According to the book, **The** Superalloys, (1) superalloys are "alloys developed for elevated temperature service, usually based on Group VIA elements where relatively severe mechanical stressing is encountered and where high surface stability is frequently required." In present commercial superalloys it is common to find niobium present in small-to-moderate amounts. These relatively small additions have a significant contributory effect on alloy properties, yet in the technical community there has not been a directioned attempt to collate and define these effects. Thus, this report **is** an attempt to accomplish this task.

Superalloys are composed of a group VIIIA metal alloy base (cobalt, nickel, or iron) to which are added large and varied amounts of alloying elements, ranging up to 50-60 percent of total alloy weight. These additions modify, improve and interact with the alloy base and with each other in a very complex fashion to provide resistance to surface attack from oxidizing and hot corroding gases, and to provide an unusual level of mechanical properties. Alloys for gas turbine service must meet stringent criteria for tensile strength and ductility, rupture and creep strength and ductility, and mechanical and low-cycle fatigue requirements, to mention only a few. Density, thermal conductivity and expansion properties are some of the physical properties usually controlled, also.

Superalloys are materials generally used to "get the most out of" highperformance equipment; they are the work horses in the hottest parts of aircraft jet engines, industrial gas turbines and very high-temperature reactor applications. Figure 1 illustrates an aircraft engine, showing a nickel-based superalloy bucket (blade) typically utilized in the turbine section of the engine. However, superalloys are also vital in high-temperature heat exchangers for energy application, in some reactors, and in a variety of other tough-service type applications.

In this paper, we will discuss cast and wrought nickel, cobalt and ironnickel superalloys and (as a special case) eutectic superalloys. In order to consider the role of niobium most efficiently, we will cover the field in the following order; Ni-base superalloys; Co-base superalloys; Fe-Ni-base superalloys; eutectics. A more detailed discussion of superalloys as a class follows.

Superalloys: Their Metallurgy and Properties

A concise definition of superalloys has been given in the introduction. In this section, an attempt will be made to review the superalloy class in a very brief fashion, — yet still crack open the door into their fascinating technological detail as background for consideration of the role of niobium. This will be done by first considering a general classification which provides a broad linking of application field with alloy type, then touching in turn on their chemical constitution, solid-state phase relations, metallurgical microstructure, processing and properties.

Classification

Superalloys have been spawned principally by the constant and driving force of the need of aircraft jet engines to operate at higher and higher temperatures for improved efficiency and power, augmented by the demands of industrial gas turbines, heat exchangers and reactors. Superalloys can be classified in many ways, one of which is given in Figure 2, a correlation of major energy systems with alloy types (2).

Chemical Compositional Characteristics and Strengthening

Superalloys are most commonly considered as either cobalt or nickel-base alloys. However, significant amounts of iron have been included with nickel to reduce cost of alloys intended for larger forgings (Fe-Ni-base alloys), although in some cases the nickel dominates; this has created an iron-base series also.

Figure 3 illustrates, in broad and qualitative fashion, the general function of alloying elements in the group VIIIA Ni-Co-Fe bases. The compositional <u>complexity</u> is obvious and Figure 4 emphasizes the point, showing that Ni-base alloy systems are the most complicated of all alloys.

Obviously, interaction between the alloying elements and alloy base, and between alloying elements also is extremely complex. However, Table I attempts some order through collation of the behavior of the major alloying element actors. One can see that mechanical properties are developed through combinations of solid solution strengthening, coherent phase precipitation,



Figure 1. Typical Aircraft Engine (GE-SNECMA CF-6) and Cast Superalloy Bucket (Insert).



Figure 2. A Correlation of Energy Technology, Major Plant Equipment, General Superalloy Type, and Specific Superalloy Examples.



Figure 3. Function of Elements Added for Alloying Effect in Austenitic Superalloys.

Fe	Fe	Fe	Zr	Ti	A1	Мо	co	Ni
NiCrMoV	M-152	31655	zy-II	3.0 AI	6061	<u>Mo-⊺ZM</u>	HS-188	TRW-VIA
3.5 NI	2.5 NI 12 Cr	12 N1 17 Cr	0.1 Fe	1.5 Fe	0.6 Si	0.08_Zr	22 Cr	6.1 Cr
0.7 Mo	2.0 Mo	2.5 Mo	0.1 Cr	1.4 Cr	0.25 Cu	0.03 C	14 W	5.4 Al
0.1 V	0.3 V	0.1 C	0.05 Ni	1.2 Mo	0.25 Cr		0.1 La	1.0 Ti
0.3 C	0.1 C						0.1 C	2.0 Mo
								5.8 W
								0.5 Cb
								9.0 Ta
								0.5 Re
								0.5 Hf
								0.02 B
								0.13 Zr
								0.13 C

Figure 4. Compositions of Various Alloy Groups, Showing Relative Complexity of Nickel-Base Superalloys.

carbide strengthening and grain boundary control. Some of the key precipitate phases formed are also identified. It should be emphasized that Table I is broad in its coverage and not intended to be totally inclusive.

Phase Relations

The structural matrix of all superalloys by definition is based on the FCC γ (austenitic) lattice characteristic of Ni-, Co-, and many Fe-base alloys. They can be considered a phase extension of the 300-Series stainless steels. Classically, their position in the austenite field ranges back and forth near its stability limits, depending upon the nickel, cobalt, or iron content and the amount of counter-austenite elements added, such as BCC chromium, molybdenum, tantalum.

Most importantly, many of these alloys are precipitation-strengthened by the coherent RCC A_3B phase γ' , formed between nickel with aluminum, titanium or niobium. The composition is usually N1₃(A1,T1,Nb), and the phase can compose up to as much as 50 percent of the alloy. Phase relations for γ' (3) are shown in Figure 5, together with a view of some typical γ' particles. Additionally, coherent phases such as γ'' (orthorhombic N1₃Nb) can form, and a broad range of carbides including the types MC, M₆C, and M₂₃C₆. The more specific behavior of some of these species will be covered in a detailed discussion involving niobium below.

A correlative picture of austenite phase behavior can be generated by viewing the polar diagram of first long period elements with chromium. A polar phase diagram is created by linking appropriate ternaries on mating axes, and specifying a common element (such as Cr) as the focus. For superalloys, the first long period elements (N1, Co, Fe, etc.) are caused to occur in periodic order at the periphery. Fairing the ternaries allows the picture to generate into a **polar** diagram (Figure 6a). It should be understood that for superalloys this diagram refers to the HCC <u>matrix</u> composition only, and not the composition of the entire alloy, including its numerous precipitate phases.

The diagram can be tailored for greater accuracy, also. For instance, if one utilizes Cr-15Mo as the focus instead of chromium alone, the related group of classic Ni-base alloys Udimet-700, Rene 77, Nimonic 115, and Udimet-500 will fall near point in Figure 6a. However, the diagram with niobium (Figure 6b) illustrates the sharp reduction in austenite field caused by element No. 41, and the complex series of phases then generated, -obviously not a beneficial effect. This will be discussed later.

	Alloying Element and Phases Formed in Designated Sy						
Metallurgical Function,	Nickel-Base	Cobalt-Base	Ni-Fe Base ^{a.}				
Solid Solution Strengthening	Co, Cr, Mo, W, Nb, Ta	Ni, Cr, Mo, W, Nb, Ta	Fe, Cr				
Coherent Phase	Al, Ti, Mo, Nb	Limited	Al, Ti, Nb				
Strengthening	Υ'		γ', γ"				
Carbide Phase Activity and	Cr, Ti, Mo, ₩, Nb, Ta	Cr, Ti, Zr, Mo, W, Nb, Ta	Cr, Mo, Nb, Ti				
Strengthening	MC, M ₆ C, M ₂₃ C ₆	мс, м ₆ с, м ₂₃ с ₆	мс, м ₂₃ с ₆				
Grain Boundary Activity	B, Zr	B, Zr	B, Zr				
Oxidation and Corrosion Resistance	Cr, Al, Ti	Cr, Al, Y, Al	Cr, Al				
Classic Alloys	Udimet 500 Inconel 738 Nimonic 80A Hastelloy X	x-40 FSX-414 MarM-509	N- 155 A-286 Inconel 718 Incoloy 903				

Table I. Summary of the Function of Alloying Elements in Superalloys.

a. Either nickel or iron may predominate.



Figure 5. N1-A1-X Phase Composition Relations, Showing Effects of Alloying Elements on Solubilities (3). Microstructure of γ Interacting with Dislocations in Nickel Alloy U-500; 10,000X.

Microstructure

Superalloy microstructure is fascinatingly illustrative of not only the strengthening from the variety of GCP and carbide phases present, but **is** the key element in successful processing, and the prime visage tool for the great developments in powder metallurgy and controlled solidification processing now occurring. Unfortunately, this brief review cannot consider all this technology so Figure 7 **is** given to show the classic features present in investment-cast alloys, and Figure 8 an example for typical forged alloys (2). The discussion below will identify additional microstructural features, particularly where niobium **is** involved.

Processing

Until about 1960, superalloys consisted of but two principle classes derived from processing, - "cast" and "wrought" - as illustrated directly previously. In contemporary practice, castings are made by the investment casting (lost wax) process often in vacuum. Highly complex shapes are possible (as in the bucket of Figure 1) and are utilized to take advantage of design factors such as air cooling in gas turbine rotating and static hardware. Wrought products are produced by a variety of working techniques to make sheet, bar, large wheels and complex parts. Precision forging is utilized for turbine blades, but complex shape needs somewhat limit this approach.

However, "cast" and "wrought" classes are no longer all-inclusive. We are now in the "Age of Processing" in superalloys. Special processing techniques are being used to extend superalloy properties to previously-unknown limits. In very brief review, some of these are:



Figure 6. Polar Phase Diagrams for First Long Period Elements VS. Cr-14Mo (left) and Niobium (right).



- <u>Powder Metallurgy</u> to obtain fine-grain ductile structures, and to create oxide dispersion strengthening (ODS).
- <u>Micrograin Processing</u> where extremely fine-grained powder is used to generate unusual structures and ductility.
- <u>Directional Solidification (DS)</u> grains elongated to the full height of the part to reduce thermal fatigue initiation. DS allows strengthening by eutectically-formed phases, and is becoming commercial.
- <u>Single Crystals</u> Wherein the entire complex part is one cast crystal. This tends to eliminate thermal fatigue failures and maximizes creep properties.
- <u>Coating and Cladding</u> to protect alloy surface attack from oxidation and hot corrosion.

Where appropriate, these processes will be explored briefly in this text.

Properties

Superalloys are characterized principally by high tensile and creep properties up to greater proportions of their melting points than all other alloy systems. In addition, they possess good-to-outstanding surface stability in simple oxidizing atmospheres. Together, these characteristics have combined to create a large group of materials which absolutely dominate hightemperature service needs up to the 1200-1300 C (2000-2200 F) surface range. Figure 9 illustrates these properties. While some refractory metals have greater strength capability, their surface stability in oxidizing atmospheres is horrendous, prohibiting practical service. Ultimately, a few precious metal alloys have greater atmospheric resistance than superalloys, but these systems are relatively weak and, of course, they are prohibitively costly. Thus, superalloy systems posses the best combination of properties. In point of fact, <u>superalloys are the only usable and durable alloys available for</u> high-stressed high-temperature oxidizing service.



Figure 9. Comparison of the Surface Stability and Strength Characteristics of Superalloys with Other Alloy Systems.

In a more broad sense, of course, specific application requires many different properties, and alloy types and classes are often tailored to these needs. Alloys for gas turbine nozzle guide-vane service for instance, require creep strength, corrosion resistance and must be weldable. Alloys for aircraft engine buckets (blades) are dominated by both thermal fatigue and tensile properties, but also must be protected from corroding atmospheres. Nuclear alloys must be free of cobalt and tungsten, but still poses great creep resistance and surface stability properties in the presence of a nuclear flux.

In order to achieve all these properties, and continue to demonstrate continued improved performance, many critical component applications now utilize two or more superalloy compositions bonded together to optimize performance. The most predominant examples are alloys of great strength protected by corrosion-resistant coatings or claddings.

Niobium: Its Properties for Use in Superalloys

In the immediately preceding discussion, it was pointed out that austenitic (FCC) superalloys are strengthened principally by addition of elements in solid solution to strain the lattice, by precipitation of finely divided coherent phases such as γ' and γ'' , and presence or precipitation of several types of carbide phases, usually incoherent. The behavior of alloying elements at grain boundaries and their effects on oxidation and corrosion resistance are also important, -often vital. Alloying elements are used to form special structures as well, such as in eutectic strengthening.

In this complex array of factors, the obvious question for us then is "What might niobium have to offer as an alloying element?"

The first consideration is a generality that hardly needs review. Niobium is a refractory metal, one of the "big four" together with molybdenum, tungsten and tantalum. All of these elements are added to one superalloy or another, singly or multiply, where they strengthen the alloys by either carbide phase formation and by solid solution strengthening or by both. With that start, a review of the properties of niobium pertinent to superalloy use follows. Niobium will be discussed with considerable reference to other refractory metals.

Periodic, Atomic and Physical Properties

Figure 10a shows niobium (columbium) in its periodic system relationships. It is apparent that niobium, positioned in Group VA, is of lower modulus and melting point than Mo, W, or Ta. It is also less dense. As a BCC metal, it forms continuous solid solutions with many other BCC metals, including 8-T1. However, for effective solid-solution strengthening, high modulus and high melting point are desirable. Since niobium has relatively low melting point and modulus (Figure 10c, Table Π), it would not be expected to be as effective as the other refractory metals under equivalent alloying conditions. Atom size (Figure 10d) is another factor in solidsolution strengthening, effecting relative solubility. Table II identifies the solubility limit of the major refractory metals in superalloy matrixes, showing niobium to be the most limited. The atomic mismatch (calculated from Figure 10) confirms that niobium creates too much mismatch to enter solution extensively. Figure 6 shows the strong effect of niobium to restrict the extent of austenite, and its great tendency to form a variety of phases with First Long Period elements.

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-3 -1 -1	4 80 30											5 8 10.82	С 2010	7 N	0.00	е Е.00	으훋렮
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19 K 39.00	2 2%	21 Sc 410	22 Ti 47.90	23 V 50.95	24 Cr 820	25 Mn 549	28 Fe 50.65	27 Co	28 NI 56.09	29 Cu	30 Zn 65.30	31 Go 91,72	32 Ge 72.00	33 As M.H	34 Se 76.94	35 Br 71.11	3 X 2
37 Rb	38 Sr	39 Y	40 Zr	4) Cb	420 20 20	43 TC	44 Ru 101.1	45 Rh	#	47 Ag	48 Cd 82.4	49 In	50 Sn	Sb Sb	62 Te 127.64	53 24.92	54 Xe 131.3
56 Cs 1323	Ba Ba	57 • Lg	77 Hi	73 Ta	74 W	75 Re	76 08	77 lr 193.1	70 P1	79 Au 87.2	음모름			83 Bi 2060	84 Po 210.0	85 A1 12101	86 Rn 222
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	• ACTII	IIVM :	SEMES	90 Th 232.12	91 Pa 231.05	92 U 2380	93 Np (237)	94 Pu 2393	95 Am (241)	96 Cm (242)	97 Bk (243)	98 Cf (245)	99 Es (253)	100 Fm (254)	101 Md (256)	102 No (254)	103 Lw (259)

Figure 10(a). Periodic System of the Elements.



Figure 10(b). Elasticity Moduli of the Three Long Periods of Elements.

Figure 10. Atomic and Physical Properties and Relationships of Niobium (Columbium) to Other Elements.



22 11 2.858	23 V. 2.617	2 4 2. G5 3	25 2.725	2 6 2 573	2 7 2.501	2 ș 2.Ni 2.Ni
40	41	42	43	44	2.685	46
3Zir9	2.00 br	2 M20	27698	2.645		2Pd5
72	73	74	75	76	77	78
3.02	₂ ∏83 4	2.10135	2 ₹35 4	050	21 709	2.769

Figure 10(d). Atomic Radii of the Three Long Periods of Elements.

Figure	10.	Atomic and	Physical Pr	coperties	and Relationships
		of Niobium	(Columbium)) to Other	Elements.

Niol	bium and Other Refr	actory Meta	ls in Aust	enitic Matr	ices.
				lement	
		Nb	Та	Mo	W
Solu at 1	<u>bility Limit w/0,</u> 000-1200 C				
IN:	Fe	2	2	3	
	Fe-20Cr	10	NA	10	2:
	со	4	5	10	1
	Co-20Cr	3	7	13	1
	Ni	10	12	26	1
	Ni-20Cr	7	12	23	3
Ato	m Size Mismatch, %				
VS:	Fe	10.8	10.6	5.7	6.3
	со	14.0	13.8	8.8	9.4
	Ni	14.7	14.4	9.4	10.
Elas	sticity Modulus, E				
	psi x 10 ⁶	15	27	50	5
	$n/m^2 \times 10^6$	100	185	345	34

 Table II. Factors Relating to Potential Solid Solution Strengthening by

 Niobium and Other Refractory Metals in Austenitic Matrices.

Chemical and Phase-Forming Properties

The periodic system (Figure 10a) shows that niobium is the most electropositive of the refractory metals. This indicates niobium could have a strong affinity for formation of A₃B-type GCP phases. Indeed, discussion of nickel alloys below will show that niobium substitutes for aluminum in γ' (Ni₃Al) in a very similar fashion as does titanium. In fact, its behavior in this respect is the most complex of any element added to superalloys. Niobium promotes at least 3 different strengthening phases (MC carbides, γ' , and γ'') and clearly is the unique driver in formation of the γ'' , the BCT Ni₃Nb strengthener in many iron-nickel forging alloys.

The electropositive nature of niobium also indicates formation of relatively stable covalent - and ionic-bonded compounds such as carbides, nitrides, and oxides, affecting in turn carbide stability, and also having relationships to oxidation behavior. Figure 11 summarizes stability behavior of these compounds.

Niobium Potential; Summary

As might be viewed by a superalloy development metallurgist from consideration of its elemental properties alone, the <u>potential</u> of niobium can be summarized as follows:

- Its moderate melting point and low modulus suggest little or no specific potential as a solid-solution strengthener. Further, its large atom diameter mismatch suggests limited solubility potential in systems of interest.
- Electropositive position in the periodic systems predicts formation of rather stable carbides and nitrides.
- However, low density (for a high-melting metal) means niobium is a desirable alloying addition for rotating components.
- o Atom size and electropositive position connotes potential solubility in established GCP phases, (i.e., Ni_4A1) and as a GCP-phase generator,
- Moderate oxide thermodynamic stability identifies that effects on alloy oxidation and corrosion resistance may be expected, with niobium competing for surface oxide formation.

Metallurgical Behavior of Niobium in Superalloy Systems

The role of niobium will be discussed by consideration of nickel-base, cobalt-base, and iron-nickel-base alloys in turn, followed by a discussion of eutectics and of its effect on oxidation and corrosion.

Nickel-Base Alloys

A significant number of conventional nickel-base superalloys contain niobium as an alloying element. Niobium is particularly apparent in nickelbase casting alloys, where about 50 percent contain about 1.0 to 2.5 percent.



Figure 11(a). Stability Relationships of Transition Metal Oxides (solid lines are those of constant standard free energy of formation from the elements).



Figure 11(c). Stability Relationships of Refractory Nitrides (room temperature).

Figure 11. Stability Maps of Niobium with Co-valent and Ionic Bonding Elements (Solid Lines Are Those of Constant Standard Free Energy of Formation from the Elements). A lesser proportion of wrought nickel alloys contain niobium, but amounts up to 6 percent are used, and its function is more broad and essential. Table III gives a representative list of these nickel alloys, together with a selection of contemporary alloys of importance that do not contain niobium, for comparison purposes.

Inspection of the alloy list does not reveal any obvious pattern (to this writer) of attention to niobium; it appears to have been utilized in a rather random manner in nickel alloy development. The objective here then is to determine the function of niobium in nickel alloys. Since strengthening is achieved principally by atom, particle, or defect interaction with dislocations through solid solution, hard particle (carbide or boride) and coherent particle effects, niobium's role will be considered for each of these latter effects in turn.

However, there are two specific phenomena of major consequence wherein niobium is a key element which must be first clarified. Niobium utilized in intermediate-temperature high-strength Fe-Ni forging alloys, such as IN-718, provokes a major strengthening effect, whereby a precipitation of γ " BCT (Ni₃Nb) occurs as a major strengthening phase. This occurs only in alloys of mixed iron and nickel content. For our purposes here, this effect is included under Fe-Ni-base alloys. Secondly, niobium plays a key role in certain eutectics, such as $\gamma \gamma' - \delta$ or CoTaC-744; that effect is also discussed separately.

<u>Solid Solution Strengthening</u>. As is well known, introduction into an alloy lattice of foreign atoms can create strain by lattice expansion which interacts with dislocations. Periodic valency effect (N) can create

strengthening effects, also. Altogether key factors related to the effectiveness of the substitute atoms are atom size mismatch with the lattice, elasticity modulus, electrical differences, and/or the effective amount of the solution element hosted by the lattice.

As summarized in Table II, niobium does not dissolve extensively in nickel, Ni-Cr or other austenitic-base systems. It is limited to the range 5-10 percent in most cases, and to ~7 percent in Ni-20 Cr at 1200 C (2200 F). Atom size mismatch, ranging from 10-15 percent, is too large to allow greater solubility. However, the large mismatch does suggest the <u>specific</u> effect per niobium atom might be significant. The periodic valence position of niobium at $N_{\rm p}$ = 5.66 suggests a strong relationship with stacking fault energy.

The behavior of niobium in Ni-base solid solution has been investigated and reported by Gua and Ma (4). Their pertinent study of the effect of niobium in a Ni-20Cr-base alloy is the first broad attempt at a definitive work. A vacuum-melted Ni-20Cr base alloy, mildly strengthened by γ' and carbon was alloyed with eight levels of niobium, from 0.5 to 2.46 w/o. The alloys were fabricated to rod and heat treated. The phases formed were separated and evaluated for niobium content, while lattice spacing and longrange order effects were studied by x-ray. Some physical and mechanical properties and microstructural studies also were reported.

Table IV summarizes completely the analytical results obtained; in connection with solution strengthening it shows the following:

o Slightly over half of the niobium concentrated in the FCC γ matrix. (Partitioning toy, γ' , and carbides is shown in Figure 12a).

Alloy (a)	C	Cr	Ni	Fe	Co	Mo		Сb	Ta	Ti	AI	В	Zr	Other	Form
Alloy 713C Alloy 713LC Astroloy B-1900 Hastelloy alloy X	0.12 0.05 0.06 0.1 0.10	12.5 12.0 15.0 8.0 21.8	Bal Bal Bal Bal Bal	 18.5	15 10 1.5	4.2 4.5 5.25 6.0 9.0	 0.1 0.6	2.0 2.0 0.1	 4.3	0.8 0.6 3.5 1.0	6.1 5.9 4.4 6.0	0.012 0.01 0.03 0.015	0.10 0.10 0.08	 	C C W C W
IN-100 IN-162 IN-738 IN-597 Inconel 625	0.18 0.12 0.17 0.05 0.05	10.0 10 16 24.5 21.5	Bal Bal Bal Bal Bal	2.5	15.0 8.5 2.0	3.0 4.0 1.75 1.5 9.0	 2.0 2.6 	1.0 0.9 1.0 3.65	2.0 1.7 	4.7 1.0 3.4 3.0 0.2	5.5 6.5 3.4 1.5 0.2	0.014 0.020 0.01 0.012	0.06 0.10 0.10 0.05	1.0V 	C C C C W
Inconel X-750 Inconel 751 MarM-200 MarM-211 MarM-246	0.04 0.05 0.15 0.15 0.15	15.5 15.5 9.0 9.0 9.0	Bal Bal Bal Bal Bal	7.0 7.0	lo 10 10	2.5 2.5	 12.5 5.5 10.0	1.0 1.0 1.8 2.7	 1.5	2.5 2.3 2.0 2.0 1.5	0.7 1.2 5.0 5.0 5.5	 0.015 0.015 0.015	0.05 0.05 0.05	 	₩ ♥ C C C
Mar M-421 MarM-432 Rene 95 SEL SEL-15	0.15 0.15 0.15 0.08 0.07	15.5 15.5 14 15 11	Bal Bal Bal Bal Bal	1.0 2.0 1.0	10 20 8.0 26 14.5	1.75 3.5 4.5 6.5	3.5 3.0 3.5 1.5	1.75 2.0 3.5 0.5	2.0 	1.75 4.3 2.5 2.4 2.5	4.25 2.8 3.5 4.4 5.4	0.015 0.015 0.01 0.015 0.015	0.05 0.05 0.05	 	C C W C C
TAZ 8A TRW 1800 TRW 1900 TRW VIA	0.125 0.09 0.11 0.13	6.0 13.0 10.3 6	Bal Bal Bal Bal		10.0 7.5	4.0 2.0	4.0 9.0 9.0 5.8	2.5 1.5 1.5 0.5	8.0 9.0	0.6 1.0 1.0	6.0 6.0 6.3 5.4	0.004 0.07 0.03 0.02	1.0 0.07 0.10 0.13	 0.5 Re, 0.4 Hf	c c c,w
Udimet 500 Udimet 700 Waspaloy B Nimonic 80A Nirnonic 115	0.08 0.07 0.07 0.06 0.15	19 15 19:5 19:5 14.5	Bal Bal Bal Bal Bal	2.0	18.0 18.5 13.5 13.2	4 5.0 4.3 3.3	 	 	 	3.0 3.5 3.0 2.4 3.7	3.0 4.4 1.4 1.4 4.9	0.005 0.025 0.006 0.004 0.16	0.07 0.06 0.04	 	C,W W W

Table III. Compositions of Selected Nickel-Base Superalloys With and Without Niobium.

Nb Content			γ' Cont	ent Wt%	Carbide Content Wt%						
in the alloys wt%	Ni	Cr	Nb	Al	Ti	Total	Ni	Ti	Nb	Cr	Total
	9.46	0.222		0.666	1.387	11.74	< 0.01	0.023		0.520	0.543
0.5 I	9.40	0.467	0.138	0.710	1.404	12.12	0.01	0.066	0.103	0.168	0.337
1.00	10.40	0.532	0.291	0.710	1.427	13.36	0.01	0.063	0.140	0.153	0.356
1.24	10.75	0.554	0.469	0.702	1.475	13.95	0.0 I	0.055	0.181	0.126	0.362
1.53	10.80	0.551	0.485	0.705	1.476	14.02	0.01	0.054	0.170	0.129	0.352
1.72	10.90	0.547	0.529	0.700	1.48 I	14.15	0.0 I	0.069	0.246	0.128	0.443
1.94	11.70	0.608	0.628	0.710	1.489	15.14	0.01	0.06 I	0.246	0.122	0.429
2.46	11.80	0.580	0.739	0.745	1.522	15.39	0.0 I	0.058	0.30 1	0.122	0.48 1
	Nb Content in the alloys wt% 0.5 I 1.00 1.24 1.53 1.72 1.94 2.46	Nb Content in the alloys wt% Ni 9.46 0.5 I 9.40 1.00 10.40 1.24 10.75 1.53 10.80 1.72 10.90 1.94 11.70 2.46 11.80	Nb Content in the alloys wt% Ni Cr 9.46 0.222 0.5 I 9.40 0.467 1.00 10.40 0.532 1.24 10.75 0.554 1.53 10.80 0.55 I 1.72 10.90 0.547 1.94 11.70 0.608 2.46 11.80 0.580	Nb Content in the alloys wt% γ' Cont 0.5 I 9.46 0.222 0.5 I 9.40 0.467 0.138 1.00 10.40 0.532 0.291 1.24 10.75 0.554 0.469 1.53 10.80 0.551 0.485 1.72 10.90 0.547 0.529 1.94 11.70 0.608 0.628 2.46 11.80 0.580 0.739	$\begin{tabular}{ c c c c c c c c c c c c c c c c c c c$	$\begin{tabular}{ c c c c c c c c c c c c c c c c c c c$	$\begin{tabular}{ c c c c c c c c c c c c c c c c c c c$	$\begin{tabular}{ c c c c c c c c c c c c c c c c c c c$	$ \begin{array}{c c c c c c c c c c c c c c c c c c c $	$ \begin{array}{c c c c c c c c c c c c c c c c c c c $	$ \begin{array}{c c c c c c c c c c c c c c c c c c c $

Table IV. MIODIUM Benavior in Experimental MI-base Austenitic Superalloys Guo and Ma	Ma (4	(4
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	Lattice	Spacing	Mismatch	Particle	Long Range	Volume Fraction	Shear
No.	<u></u>	<u>α</u> Υ	ε, %	Radjus r, a	Order Parameter, S	of Y'	Modulus G
					,_		3
1	3.5906	3.5634	0.76	84.5	0.79	12.56	81.7 x 10
2	3.5933	3.5666	0.75	85.5	0.82	12.98	82.4 x 10 ²
3	3.5951	3.567 1	0.78	98.5	0.85	14.18	82.8 x 10
4	3.5965	3.5674	0.8 I	101.5	0.86	14.85	83.8 x 10 ²
5	3.5967	3.5682	0.80	98.5	0.87	14.84	83.8 x 105
6	3.5970	3.5701	0.75	98.5	0.92	14.95	84.2 x 10 ²
7	3.5976	3.5701	0.77	105	0.95	15.91	2
8	3.5987	3.57 13	0.77	113	1.01	16.21	85.1 x 10 ⁹



Figure 12. Partitioning of Niobium toy, γ ', and Carbides in Ni-base Alloys with 0-3.0% Niobium. After Gua and Mi (4).

- \circ Niobium increased the lattice spacing of γ slightly.
- **o** Lattice mismatch of γ to γ' increased from 0.76 at 0 percent Nb to 0.81 at 1.24 percent Nb; the mismatch then declined to nearly 0.76 at 2.5 percent Nb.
- Alloy shear modulus increased from 81.7 x 10^3 N/mm² to 85.0 x 10^3 , -an increase of about 4 percent.

From these data, it is obvious that niobium has a definite solution strengthening effect, and also generates significant mismatch effects between the matrix and γ' . The authors estimated lattice distortion should increase the

yield strength by 44 N/mm^2 (~ 6,300 psi), one-third of the total yield strength gained. Electronic effects were not discussed.

The question still remains as to whether useable solid strengthening effects continue up to the solubility limit of about 7 percent Nb, - but it may be a moot point. The most useful effect from niobium is in carbide and coherent phase strengthening, as discussed subsequently, and effects in solution relate principally to effect on coherent phase mismatch.

<u>Carbide Strengthening</u>. Depending upon carbon and alloy content, a relatively complex series of carbides can be found in nickel superalloys. Their composition varies over a range from MC (such as TiC) through M_2^C and $M_{23}^C_6$ to M_6^C . These carbides react constantly with each other and with the alloy matrix in heat treatment and in service. They provide a multitude of useful services in the alloys, including at least:

- Structure refinement during cooling from heat treatment or during fabrication.
- Grain strengthening through matrix precipitation at structural aspersions.
- o Grain boundary "locking" to prevent slip.
- o General dislocation generation.

Among the carbides, the MC form is usually found profusely in early alloy history. During alloy thermal exposure the MC usually degenerates according to the reaction (5):

$$MC + \gamma + M_{23}C_6 + \gamma'$$
(1)

The $M_{23}C_6$ can be located at grain boundaries, sometimes with a strong tough layer of $\gamma^{\,\prime}$ formed over it. An example of this reaction history in alloy 713C (2.0% Nb) is shown in Figure 13, taken from the classic work of Mihalisin, et al (6).

Niobium is important in this reaction, since it partitions to the MC carbides in favored fashion and is believed to be instrumental in helping control the stability (i.e., - reduces reaction rate) of the MC. Decker and Bieber (7) have analyzed MC carbide in Inco-713C to be $(Nb_{0.40}^{-Ti}_{0.09}cr_{0.33}^{-C}_{0.33})^{C}$, showing the dominant role niobium can play. Beattle (9) has identified that the stability of MC carbides varies as follows:



- a. As Cast; MC Carbide Particle
- Figure 13. Decomposition of NbC in Alloy 713C (2.0% Nb) During Thermal Exposure, Showing Generation of M₂₃C₆ at Grain Boundaries. All Structures at about 10,000X. Adapted from Studies by Mihalisin, et al. (6).



b. Alloy Exposed at 1040°C (1900F) for 1000 Hrs. Showing Near-Complete MC Solution, with Surrounding $M_{23}C_6$ and γ' Formed



c. $M_{23}C_6$ and γ' Formed at Grain Boundary

Also, when molybdenum enters the MC structure, -viz. (Ti, Mo)C, -it decreases the stability of the TiC. Conversely, it is generally held that both tantalum and niobium stabilize MC carbide, decreasing the tendency to pursue reaction (Equation 1). This latter fact was first mentioned by Harris and Child (9) who in a detailed study of carbide-forming elements in Fe-Ni-base alloys (discussed subsequently) concluded that niobium not only formed profuse stable carbides but singularly depressed formation of chromium carbides (1.e., $M_{23}C_6$). Further, Lund (10) reports that niobium appears to stabilize MC more strongly than tantalum. In development of alloys such as MM-200, niobium created such a stable MC carbide, viz., (Nb, Ti)C, - that reaction (Equation 1) could not proceed at all. In these cases, tantalum was added to reduce the MC stability, since (properly controlled) the reaction is considered essential for many alloys.

Certain other high-niobium alloys (such as MarM-421 and Inco-713C) show pronounced degeneration of MC (7), while some non-niobium alloys (IN-792) generate very stable MC carbides. In those cases where NbC does form and later decompose, it is highly possible that, like titanium, the niobium helps generate more γ ' in the alloy through reaction (Equation 1), which can be written to include niobium approximately as follows:

NbC + (N1,Cr,A1) +
$$Cr_{23}C_6 + Ni_3Nb$$
 (2)

In the report of Gua and Ma (4) 15-20 percent of the niobium (Table IV) partitioned to the carbide phase. While the specific type of carbide was, unfortunately, not reported, its composition suggests an MC structure.

<u>Coherent Phase Strengthening</u>. As discussed near the beginning of this paper, the most significant single metallurgical strengthening phenomenon in superalloys is precipitation strengthening from the FCC-GCP phase γ' , based on the composition Ni₂AL. Microstructural views of two alloys with γ' containing niobium are shown in Figure 14 (see also Figure 13). Gamma prime causes strengthening through generation of coherency strains with the lattice, and effects APB energy in dislocation cutting; strength of the γ' and the γ , size of the γ' forms a broad range of composition and accepts elements such as chromium and cobalt in substitution for the nickel, and titanium and niobium for the aluminum.

Superalloy metallurgists have found that additions of titanium and niobium to alloys increase the amount of γ' (there is always sufficient nickel) and change its stability characteristics as well. Guard and Westbrook (3) (Figure 5) have shown the liberal amount of titanium and niobium which can substitute for aluminum. In fact, one can substitute titanium for nearly 75 percent of the aluminum, still retaining γ' . Niobium also dissolves extensively in γ' , increasing the γ' volume fraction in alloys such as SEL-15, MarM-200, IN-713C and probably increases γ' solution temperatures as well (12). Mihalisin (7) analyzedy' in 713C (Figure 13) showing the following composition (Ni.98^{Cr}.02^M.004⁾3^{(A1}.71^{Nb}.10^{T1}.05^{Mo}.04^{Cr}..10). Thus, niobium accounted for about 10 percent of the y' precipitate on an atom concentration basis.





Figure 14(a). Alloy MarM421 (2.3% Nb). F Heated at 87 C (1600F) for 1000 Hours. 4250X. After Ault and Donachie (11).

Figure 14(b). Alloy SM-200 (2.0% Nb). As cast. 5500X. After Lund.

Figure 14. Examples of y' in Ni-base Alloys Containing Niobium.

Evaluation of a series of superalloys (13) showed niobium could reach 3 percent of the whole (12% of Al). While niobium did not increase γ ' hardness at room temperature, it could be a significant strengthener at high temperatures (3). Further, Thornton, et al (14) have demonstrated specifically (Figure 15) that niobium (and titanium) doubled the flow strength of γ ' itself at useful temperatures, -600-900 C (1100-1650 F).

The study of Gua and Ma is further instructive here, also (Table IV) (4). They found that:

- o Niobium concentrated in they' in approximately 2:1 ratio to the matrix
- o As niobium content in the alloy increased from 0 to 2.5 percent, the amount of γ ' increased 3.7 percent n the alloy, and the volume fraction of γ ' increased 30 percent, from 12.56 to 16.21.
- o Lattice mismatch withy increased to 0.8, then decreased to nearoriginal at 0.77.

Thus, niobium significantly increases the amount of γ' . Further, the niobium in the alloy matrix decreased aluminum and titanium solubility, allowing still more γ' to form.

Gua and **M** also evaluated the effect of niobium on the coherent strain field, and pointed out that while the YS increment due to coherent strain is large, lattice mismatch change caused by the niobium is not significant in increasing strengthening through coherent strain. However, utilizing the value of long-range order determined, - which increased with increasing niobium, - the APB energy was calculated and found to increase by about 60 percent. A yield strength increment calculation showed up to about half of the gross alloy yield strength factor was due to this factor.

<u>Nickel Alloy Summary</u>. The significant strengthening effects of niobium in nickel-base superalloys appear to be as follows:



Figure 15. Effect of Alloying on Flow Stress of γ '. After Thornton, et al. (14).

- o Niobium concentrates iny', increasing volume fraction significantly
- In γ', niobium increases the APB energy, increasing resistance to dislocation cutting, - and thus increases high-temperature strength.
- o Niobium reacts with carbon to form a very stable MC; it thus **can** be utilized to control the MC degeneration reaction (Equation 1).

Cobalt-Base Alloys

Niobium has been utilized as a strengthener in several commercial cobaltbase superalloys. Analysis of its behavior is relatively simple compared to that in nickel or Ni-Fe-base alloys, since coherent GCP-type phases (such as Ni₃Al) are not formed usefully. In commercial cobalt superalloy use, it

appears that niobium's role is principally that of only solid-solution strengthener or a carbide-former. Nonetheless, niobium has been studied intensively as a potential generator of GCP-type phases, as will be discussed below.

Table V identifies several cobalt-base alloys containing niobium, as well as several that do not; the latter are for baseline comparison. While not extensive in variety, 10,000-20,000 tons of these alloys are utilized annually in industrial gas turbines, aircraft engines, furnace components, and other applications. X-40, WI-52, and FSX-414 are among the most voluminous of all superalloys cast, and S-816 one of the earliest. Historically, investment-cast cobalt superalloys were the first successful American jet engine alloys.

In general, these alloys are cast readily in air or vacuum to complex shapes, possess useful creep strength to higher temperatures than most conventional nickel-base alloys, and offer excellent hot-corrosion resistance. Niobium content ranges from about 2.0 percent to 4.0 percent in typical alloys. The behavior of niobium will be discussed by strengthening function below.

<u>Solid-Solution Strengthening</u>. Most of the cobalt-base alloys contain several types of refractory metals. These form stable carbides early in the alloy freezing process, but are sufficiently in excess for significant solidsolution strengthening effects also to be apparent. An important exception may be the alloy UMCo-51, in which most of the niobium is present only as a carbide.

Detailed examination of the behavior of niobium in solid-solution strengthening appears limited. Solubility limits in Co-Cr alloys appear to be about 3 percent at 1000-1200 C (1800-2200 F), Table II. This is the lowest of the group Mo, W, Ta, and Nb. Nb-Co atom size mismatch is about 14 percent, much too great for extensive solutioning (Figure 11). Further, the elastic modulus (Figure 10) is low, - about 14 x 10^6 compared to 30 x 10^6 for cobalt. All of these three factors are sufficient to predict poor solution-strengthening effects. In fact, molybdenum and tungsten (and even tantalum) are much more effective solution strengtheners in cobalt. Niobium has a much more significant role in these alloys as a carbide-forming element.

<u>Carbide Strengthening</u>. As in other austenitic systems, niobium combines to form strong, stable carbides in cobalt. A prime example is S-816, where niobium is the principal element reacting to form the primary carbide, NbC; NbC in S-816 is stable up to 1200 C (2200 F), where it is only very slightly soluble. An intermediate carbide, of the M_{c} C type is formed also, and

	Composition, wt%											
Name or Tradename	<u> </u>	Cr	<u>Ni</u>	Mo	W	Nb	Ta	Ti	<u> </u>	Fe	Other	Alloy Type
S-816	0.38	20	20	4	4	4				4		Wrought
WI-52	0.45	21	-		11	2				2	0.25 Mn	Cast
UM Co-51	0.3	28				2				20		Cast
AiResist 13	0.45	21			11	2				2.5	0.1 Y	Cast
X-40	0.50	25	10		7.5					1.5	0.75 Mn	Cast
FSX-414	0.25	29	10		7.5				0.010		0.75 Min	Cast
L-605	0.10	20	10		15							Wrought
MarM-509	0.60	24	10		7		7.5	0.2		Ι		Cast

Table V.	Compositions	of	Cobalt-Base	Superalloys	With	and	Without	Niobium.

appears to be principally $(Cr,Mo)_6C$ in nature. Since MC carbide in the alloy S-816 is a major strengthener and source of carbon for M_6C , niobium appears to be the most significant alloy strengthener. Effects of niobium in WI-52 and AiResist 13, - while not identical, - are quite similar. All of this behavior is consistent with the observations of carbide stability by Beattie (8) discussed above.

It should be clarified that in such carbide-strengthened cobalt alloys, the MC can be large and blocky and/or "Chinese Script" in shape. One school of thought feels that niobium promotes the blocky type, which is favored for alloy ductility. Both MC carbides function principally as stable, thermally-resistant sumps of carbon. During heat treatment and service they gradually enter solution, with the carbon reprecipitating to form finely-divided lower carbides such as $M_{6}C$ and $M_{23}C_{6}$ by reaction with chromium and other matrix

elements. These carbides can be redissolved solutioned and precipitated. The stable, relatively coarse NbC also acts as a grain refiner during cooling of castings (it can precipitate in the melt prior to casting) and to refine structure during forging, - or during cooling from heat treatments. Typical structures of these carbides are shown in Figure 16.



Figure 16. Structures Typical of Cobalt-Base Superalloys Containing Niobium. a. Massive NbC Particle in S-816. 10,000X. After Beattie

b. General Structure Showing Angular MC Carbides, with Eutectic Areas $(M_{23}C_6)$ which Tend to be Devoid of Niobium. 500%.

An important application of Co-base alloys is for heavy-duty furnace parts, such as runners and tubes in the metallurgical, chemical and petrochemical and petrochemical industries, where oxidation and corrosion resistance, erosion resistance, and strength are needed. UMCO-50 (50 Co, 28 Cr, 21 Fe, 0.3 C) has been utilized for many years in this service. Ultimately a niobium-containing version, UMCO-51 (Table V) was issued (16). In studies to improve the UMCO alloys, it was found that additions of niobium systematically improved the strength through interaction to form NbC-type carbides in addition to the $M_{23}C_6$ carbides already present. In fact, aging studies for 3000 hours up to 800 C (1470 F) showed increases in strength, but with no tendency towards unwanted a-phase formation, since the Phacomp phase control tool was utilized to control composition. Figure 17 shows the carbide structure obtained in one alloy, and illustrates the effects noted from systematic increases in niobium content. However, maximum properties (improvement in both UMCO-50 and UMCO-51) are obtained by concurrently replacing a moderate amount of the cobalt with nickel. These NbC-type carbides probably also contain nitrogen, - becoming carbonitrides.

Such alloys, as noted previously, are considered a good example of the necessity for creating a major carbide reservoir by adding Nb to form initially stable and relatively coarse NbC, - to subsequently degenerate slowly in service. The constant supply of carbon in turn, generates the more pervasive $M_{23}C_6$, which constantly dissolves and reprecipitates at grain boundaries, MC/matrix interfaces, stacking faults, and other structural aspersions acting as a major strengthener through $M_{23}C_6/d$ is location interaction. The niobium, in this act of the play, will go back into solid solution.

<u>Coherent Phase Strengthening</u>. The success found in using coherent phases to strengthen Ni-base alloys (e.g., $-\gamma'$, N1₃Al) has been one of the most strikingly useful phenomena discovered in all metallurgy. Thus, for years alloy developers have hoped for a similar effect in cobalt alloys. However, the analog to Ni₃Al, Co₃Al per se, is not stable. A most complete search was conducted by Drapier, et al (15) who evaluated the nature and morphology of intermetallic compounds formed between cobalt and the refractory metals, – niobium, tantalum, molybdenum, and tungsten.



Figure 17. Effect of Niobium on the Structure of UMCo-50. 500X.

In this study, one cubic phase, $\alpha - \cos_3 Ta$, was found to precipitate (in Co-Cr alloys) in an ordered coherent (spinodal) fashion akin toy' (Ni₃Al). However, $a - \cos_3 Ta$ is metastable, transforming to $a\beta - \cos_3 Ta$ hexagonal form, and, has not appeared useful. $\cos_3 Ti$ also forms, but it is not stable at high temperatures.

For niobium, a summary of the possible intermetallic components is reported in Table VI, showing three types of Laves phases, $-Co_2Nb$, Co_3Nb , and possibly Co_5Nb_2 . Some age-hardening tendencies were observed by precipitation of these intermetallics in Co-base binary alloys with from 5 to 25 percent chromium in 5 percent increments. Alloys with tantalum, molybdenum, and tungsten were studied similarly.

Age-hardening effects were found in Co-Cr-Nb alloys at moderate temperatures, due to generation of a superstructure in the cubic matrix, and the hexagonal Laves phase Co_2Nb (MgZn₂-type) in the HCP phase. At high temperatures (1200 C) the hexagonal MgNi₂-type Laves phase occurred, apparently stabilized by the chromium. From all of this work the authors concluded that while some hardening occurred by addition of niobium to cobalt, overaging occurs at temperatures too low to be αf interest, and tantalum, tungsten, and molybdenum generated much more provocative results. Figure 18 shows the strengthening effects of the refractory metals added to a Co-Cr base, and illustrates well that niobium has but poor potential as a strengthener through coherent phase precipitation. However, niobium was not investigated as a partial substitute for tantalum.

<u>Cobalt Alloy Summary</u>. Niobium has been explored rather extensively for possible use as a solid solution or coherent phase-former for strengthening Co-base alloys, and found, to date, not to be helpful in these modes. Its low modulus and limited solubility is a major factor, and A_3B -type phases are not thermally stable.

However, as for nickel alloys, niobium forms stable and refractory carbides in cobalt superalloys. These are highly effective strengtheners in widely-used commercial alloys, such as S-816, WI-52 and UMCO-51.

Iron-Nickel Alloys

For the purposes of this paper, this section refers to alloys which are based on iron (Table VII). Virtually all of these alloys contain a significant amount of nickel and in fact a few such as IN-718 and Pyromet 860 contain more nickel than iron. However, all of these alloys are characterized by metallurgical features and use patterns related to the presence of large amounts of iron, **so** they are chosen to be discussed as iron-nickel-base alloys. There are several generalities concerning Fe-Ni alloys that should be made at this point to clarify and shorten subsequent discussion.

- The Fe-Ni-type alloys principally are used for forged or wrought structural applications to about 650 C (1400 F). Some are used for low-strength casting applications to higher temperatures.
- o Iron-nickel superalloys utilize niobium for strengthening more than do any other class. In some Fe-Ni alloys niobium can be called the <u>principal</u> strengthener.

	%	CO			Lattice Parameters,
_Compound	a/o	w/o	Stability	Structure	a
	74.5-75.2	65-67.3	up to 1175 C (?)	hexae , Laves MgNi ₂ type	a = 4.740 c = 15.45 c/a = 3.259
3,	73	63.3	?		a = 4.738 c = 15.46 c/a = 3.262
	73	63.3	*		a = 4.75 c = 15.56 c/a = 3.27
Co ₅ Nb ₂ (?)	71.5 (?)	61.4 (?)	?	hexagonal	a = 5.89 c = 6.83 c/a = 1.16
Co ₂ Nb	67 . 4-72.7 66.7 66.7	56 . 8-62.7 56.0 56.0	up to 1175 C ? ?	cubic, Laves, MgCu ₂ type	a = 6.713–6.770 a ≡ 6:769
Co ₂ Nb	66.7	56.0	**	hexag., Laves, MgZn ₂ type	a = 5.190 c = 8.384 c/a = 1.615

Table VI. Intermetallic Compounds in the Co-Nb System. After Drapier (15).

*As stabilized by Cr ****As** stabilized by Ni.



Figure 18. Strengthening Effects of Nb, Ta, Mo, and W in Co-base Alloys with Appropriate Chromium Levels. Adapted from Drapier, et al. (15).

Alloy	С	Cr	Ni	Fe	Co	Ti	AI	В	Мо	W	Nb	Other	Form
A-286	0.05	15.0	26.0	Bal		2.00	0.20	0.005	1.25			0.3 V	w
v-57	0.05	15.0	25.5	Bal		3.0	0.25	0.008	1.25			0.3 V	W
Unitemp 212	0.08	16.0	25.0	Bal		4.00	0.15	0.060			0.50	0.05 Zr	W
Inco-901	0.05	13.5	42.7	34.0		2.50	0.25	0.015	6.1				W
Pyromet 860	0.05	14.0	42.5	Bal	4.0	3.00	1.25	0.010	6.0				W
IN-718	0.04	19.0	52.5	Bal		0.90	0.50	0.005	3.05		5.30		W.C
IN-706	0.02	16.0	40.0	Bal		1.70	0.30	0.004			2.75		Ŵ
Inco-903A	0 I		37.5	Bal	14.0	1.6	0.02				5.0		W
Rene 62	0.05	15.0	Bal	22.0		2.50	1.25	0.005	9.0		2.25		W
Udimet 630	0.04	17.0	Bal	18.0		1.00	0.60	0.005	3.0	3.0	6.50		W
D-979	0.05	15.0	45.0	Bal		3.00	1.00	0.010	4.0	4.0			W
CG-27	0.05	13.0	38.0	Bal		2.50	1.50	0.010	5.5		0.6		W
H-155	0.15	21.0	20.0	Bal	20.0				3.0	2.5	1.0		W.C
S-590	0.43	20.5	20.0	Bal	20.0				4.0	4.0	4.0		Ŵ
CRM-180	0.75	23.0	50	Bal	5.0			0.003	1.0	1.0	2.0	5.0 Mn	W
G-18B	0.40	13.0	13.0	Bal					2.0	2.5	3.0		Ŵ

Table VII.	Compositions	of	Selected	Iron-Nickel	Su	peralloys	With	and	Without	Niobium.
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- Early alloys, such as S-590 and N-155, are strengthened only by solution effects and carbide precipitation. In this respect they are similar to cobalt-base alloys.
- o Most significantly, alloys such as IN-718 show generation of a unique and vital coherent strengthening phase which depends upon niobium, called γ ".
- o The amount of nickel in the austenite matrix is critical. About 25 percent is required to maintain the FCC structure and allow precipita-tion of γ' .

<u>Solution Strengthening</u>. As with nickel- and cobalt-base alloys, solution strengthening effects occur from additions of cobalt, chromium, molybdenum and other refractory metals. The behavior of niobium as a solution strength-ener was discussed in some detail for nickel-base alloys above. In recent measurements (17) a heat of IN-718 (nominally 5.1% Nb) was found to retain about 3 percent Nb in the austenite matrix, - about the same as for nickel alloys according to Stoloff (18). Thus, the solution strengthening effects are probably similar. Since $\gamma \gamma'$ imisfit is not an important source of strengthening, effects of niobium in solution on this factor probably can be ignored. The role of niobium in the precipitate phases themselves is far more significant, as follows.

<u>Carbide Strengthening</u>. The iron-nickel alloys all contain MC carbides, formed principally with titanium but complexed by niobium and other refractory metals. Since most of the alloys are forged and MC carbides form at very high temperature, they can be vital in structural control such as grain refinement during fabrication or grain control in heat treatment. This is particularly important as forged alloys are used for wheels and other large parts. Niobium helps maintain stability of the MC carbides, but it still degenerates to $M_{23}C_6$ or M_6C carbides in the same fashion as for nickel-base alloys, particularly during heat treatment. The specific partitioning of niobium to the carbides appears not to have been studied, but if behavior is similar to nickel alloys, one can expect about 10 percent in MC.

Among the Nb-containing Fe-Ni alloys (Table VII) H-155, S-590, and CRM-180 precipitate carbides as their principal strengthening phases. The high carbon level for this purpose is obvious. MC-type carbides, composed of niobium, molybdenum, tungsten and chromium dominate; they precipitate at grain boundaries, and (as 'in G-18B (9)) on matrix dislocations as well as other aspersions in the alloy. MC carbides also are important in alloys strengthened withy ' and y", where they help control grain structure.

<u>Coherent Phase Strengthening</u>. Many Fe-Ni superalloys are characterized uniquely by precipitation of two coherent phases, γ' and γ'' . To discuss this behavior the two phases each will be discussed briefly, then covered in detail by discussion of IN-718 as a leading example.

Ordered and coherent gamma prime (γ ') is formed in Fe-Ni-base alloys by reaction of <u>titanium</u> with nickel, not primarily aluminum as in nickel-base superalloys, although aluminum participates if present. γ ' precipitation usually occurs as spherical particles. Strengthening mechanisms utilizing γ ' have been well summarized by Muzyka (19) who reports that particle size and volume fraction are significant, while matrix coherency relationships are not; it has also been observed (20) that APB energy contribution is about 10-20 percent. Further, a major function of \mathbf{y} ' is its contribution in converting to \mathbf{y} ".

Gamma double prime (γ ") is a prime strengthener in several of the most important alloys. While identified by a variety of workers (20, 21) some still contest its existence. In any case, γ " is a BCT-type phase (like two stacked FCC γ ' cells). It forms as discs or platelets in the alloy. It uniquely depends on the presence of both niobium and iron for formation; they provide needed electron-to-atom ratios and matrix-precipitation mis-match effects. Figure 11 is a summary of leading work showing the approximate quaternary phase relations between γ , γ ', γ ", and other neighboring phases.

Gamma double prime has been observed to form by enveloping γ' cubes on all sides (21). Once formed, it is more stable than γ' . In turn, the γ'' eventually may convert to the most stable phase in this series, $Ni_2Nb(\delta)$.

Thus, depending strongly on specific alloy chemistry and heat treatment a series of irreversible (without heat treatment) reactions giving differing phases and effects can be generated (Figure 19).

$$\gamma^{*}(FCC) \rightarrow \gamma^{*}(BCT) \rightarrow 6 \text{ (Orthorhombic)}$$
(3)

n (HCP)

(4)

These alloys must be melted, fabricated, and heat-treated with care!

<u>Non-Coherent Phases</u>. Delta phase (δ) is not coherent with the matrix, and forms as plates or cells. Generally, delta does not contribute to strength and thus is to be avoided. However, it has been found possible to create overaged delta (globular or platelike) by Eiselstein (23) which can control grain size during hot working.

Eta phase $(\eta - \text{Table VIII})$ can form directly from γ' in these alloys, due also, of course, to the high levels of titanium and niobium. Characteristically, it will occur as plates or cells, the plates often growing from grain boundaries where it reduces ductility leading to undesirable properties. However, careful processing and heat treatment can cause a more blocky form to generate from cellular structure, useful in grain control in a similar fashion to 6. Since both η and δ contain niobium and titanium, they can reduce strength by robbing γ' and γ ," reducing the amount present and reducing stability.

<u>TCP Phases</u>. In a similar fashion to the formation of σ or μ in the Nibase alloys, the Ni-Fe alloys can form TCP phases and Laves. Laves is a common problem, related again to the presence of niobium as well as aluminum and silicon. σ has also been reported in these alloys, but the role of niobium is secondary here, since reduction of matrix nickel (by A_3B precipitation) is a major driver to form σ along with chromium and molybdenum. TCP phases form in areas of high segregation of the forming elements, *so* good homogenization practice can avert this problem.

In summary, as listed by Muzyka (19) the most important precipitate non-carbide phases in Fe-Ni alloys are:



Figure 19. Sketch of Approximate Relationships Between γ ' and γ '/ δ in the Ni-Al-Cr-Nb System.

Table VIII. Superalloy Phases Containing Niobium.

Name	Symbol	Nominal Chemical Formula	Crystal Type	Estimated Niobium Content
Gamma c Austenite	Y	Ni,Co,Fe Solid Solution	FCC	upto5%
Gamma Prime	Υ'	Ni ₃ (AI,Ti,Nb)	Ordered FCC	Up to 25%
Gamma Double Prime	Y''	Ni ₃ (Nb,Al,Ti)	Ordered BCT	Up to 25%
Delta	δ	Ni ₃ (Nb ₈ Ti ₂)	Orthorhombic	Up to 25%
Eta	n	Ni ₃ (Ti,Nb)	НСР	?
MC Carbide	MC	NbC	Cubic	Up to 50%
M ₆ C Carbide	M ₆ C	(Nb,Mo,Ni) ₆ C	Complex Cubic	?
Laves		(Fe,Cr) ₂ (Ti,Nb)	Hexagonal ^{MgZn} 2	v28%

.

Symbol	Structure	Composition		
γ'	Ordered FCC	№i ₃ (A1, Ti)		
Υ"	Ordered BCT	NI 3NP		
Ni ₃ Ti (or ŋ)	HCP	Ni ₃ Ti		
Ni ₃ Nb (or b)	Orthorhombic	NI 3ND		

Behavior of IN-718. Alloy IN-718 is the industrially prominent example of $\mathcal{R}e-Ni$ alloys which demonstrate the principles identified above; it was developed by &. L. Eiselstein (\mathcal{Z}_0). While alloys such as Pyromet 860, Rene 62 and Udimet 630 also demonstrate some aspects of this behavior, IN-718 dominates its present field of commercial application with an absoluteness not enjoyed by any single nickel- or cobalt-base alloy in theirs. It contains about 5 percent niobium as the most prominent strengthening addition, and currently is being developed for use in cast components in addition to its historically forged condition. Many studies have been made of the alloy. Here we will attempt to be brief.

IN-718 (Table VII) is Ni-rich with about 20 percent iron. It is used for forged turbine wheels, shafts, supports and cast frame sections in aircraft engines. A major effort in all forging or casting is to process the alloy to avoid segregation which could cause formation of deleterious Laves, μ or **a**.

In forging practice IN-718 is usually fabricated by furnace heating to about 1100 C (2025 F) then rolled or forged, maintaining minimum stock temperatures of about 925 C (1700 F). The alloy is first then heat-treated by solutioning at about 950-1000 C (1750-1840 F) then aged at about 720 C (1325 F), furnace cooled to 620 C (1150 F) then cooled to room temperature. This practice avoids Laves and σ formation, generates γ' and γ'' in optimum quantities to provide maximum strength and toughness, and minimizes subsequent transformation to δ phase.

Rizzo and Buzzanell (27) showed that variations in niobium content in IN-718 had the most significant effect of all alloying elements. Their results (Figure 20) showed found steady increases in strength by increasing niobium from 3.5 to about 6.5 percent. However, additions of niobium above about 5 percent promote Laves phase and orthorhombic Ni₃Nb (δ), - both deleterious to alloy toughness.

The microstructure of forged IN-718 is shown in Figure 21. A typical view of the very fine A_3^B phases appear in Figures 21a and b Figure 21c shows typical phases generated by coarsening the structure.

Cast versions of IN-718 are now being developed following successful use in aircraft engine frames. Cast or ingot IN-718 has been unfortunately characterized by visible segregation "freckles", now identified as Laves phase and porosity. This inhomogeneity affects properties seriously, since the Nb (28%) and Mo (12%) locked into Laves is not then utilized for strengthening. It is possible (although difficult) to homogenize the alloy by heat treatment alone; Bouse and Schilke (17) found that a combination of hot isostatic pressing (HIP) and heat treatment (i.e., - four hours at 1213 C (2217 F) and 103 MPa, (15 ksi)) were effective in lowering the Laves to undetectable limits. This resulted in 600 C (110 F) increases of 0.2 percent from 780 to 920 MPa (105-125 ksi) with good retained ductility.



 Yo
 Cb

 Figure 20. Effect of Nb Variations on Ni₃Nb (γ.") Level and of IN-718. After Rizzo and Buzzanell (27). Solutioned at 950 C (1750 F) and Aged 1500 hrs. at 650 C (1200 F).



Transformation of γ' to $Ni_XNb (\gamma'')$ with Grain-Boundary Ni_3Nb Plates. 500 hrs. at 705 C (1300 F). 7500X.



Figure 21(c). SCM of Large NbC Particle (center) and Grain-Boundary Laves. 5000X.

Figure 21. Some Microstructures Typical of IN-718. After Boesch (24), Stroup (28), and Biss (29). Finally for IN-718, a summary of many studies of T-T-T relationships are given in Figure 22. It serves to summarize this brief view of IN-718 and re-emphasizes the role of niobium. The messages from Figure 22 are:

- MC (i.e., (Nb,Ti)C) is stable at high temperatures and yields to M₆C at longer times. However, both carbides are relatively unstable at service temperatures, and will be entering into chemical reactions.
- o Generation of sufficient γ " to provide good strengthening demands relatively high niobium content, such that (Ti + A1)/Nb = 0.96. At lower Nb levels (ratio = 0.77) formation of γ " is **much** more sluggish.
- o Laves will form only if the alloy is highly segregated.

<u>Other Alloys</u>. Our discussion has emphasized IN-718 for the sake of brevity. However, bothy' and γ " occur in several commercial alloys with greater iron content, such as IN-706 and IN-903A (Table VII), - a low-expansion wheel alloy. In fact, some views of γ ' and γ " in the latter alloy, resulting from a study by Bricknell and Woodford (33) are most clarifying



Figure 22. Time-Temperature-Transformation Diagram for Alloy 718 after Annealing in the Range 2100-2000 F/l hr. and Water Quenched. Laves is a non-equilibrium phase that may appear in niobium-segregated areas.

Phase Transformations Determined as Follows:

 γ ', MC, and M₆C - Eiselstein (23) γ '' - Cozar and Pineau (21) δ , Ni₃Cb - Boesch and Canada (24) Laves - Keiser and Brown (25) (Figure 2). γ " appeared after 5.5 percent Nb was added, and a heat treatment at 744 C (1375 F) 8 hrs, FC to 621 C (1150 F)/8 hrs was given. The cubic γ ' in Figure 23a is obvious, with tiny plates of γ " evident between and long plates of 6-Ni₃Nb traversing the photograph. Figure 23b reconfirms the

presence of \mathbf{y} ", showing another reflection. However, it is important to note that in this particular alloy, γ " is not believed the principal strengthener; it isy'.

<u>Iron-Nickel Alloy Summary</u>. Niobium is a vital strengthener in several Ni-Fe-base superalloys, such as IN-718 and IN-706. In the presence of iron, a new coherent phase γ " is formed at the Ni₃Nb composition, which evokes

outstanding strength for use up to about 650 C (1200 F). Niobium as Ni₃Nb also forms delta phase (δ), which does not contribute to strengthening significantly.

Directionally Solidified Eutectic Alloys

Through use of the directional solidification process, there have been recent significant American and French developments to produce <code>maturally-aligned</code> fibrous or lamellar structures of superalloy materials with high <code>creep/rupture</code> capability. The structure is achieved by freezing out a stable <code>eutectic</code> compound. Typically in a nickel-or cobalt-base alloy matrix, fibers (needles) of tantalum carbide (TaC) or niobium carbide (NbC) or lamellae (plates) of orthogonal Ni_aNb (δ) are formed. In the case of nickel alloys,

 γ ' is usually also precipitated, and other carbides can form in most of the alloys.

Table IX gives a listing of some major eutectic alloys. In the general sense, these are not commercially available yet, but developments are now reaching towards at least extended trial service. For instance Brown-Boveri company has tested a chromium carbide fiber alloy in industrial gas turbine service for about 15000 hours, (30) and the pressure is on for aircraft engine application.

The leading alloys at present appear generally to be of the following type:

- NITAC. A TaC fiber structure in an austenitic nickel superalloy matrix.
- o $\underline{\gamma} / \underline{\gamma} \underline{\delta}$. A lamellar $\aleph_{1_3} \aleph_{\delta}(\delta)$ structure in a nickel γ '-strengthened alloy.
- <u>COTAC-744</u>. An NbC fiber structure in a <u>nickel</u> γ¹-strengthened alloy. (Note: this alloy name is strikingly misleading, - since the word COTAC historically refers to TaC fiber <u>cobalt</u>-base alloys, with matrixes similar to X-40, on which much early work occurred).

The significant role of niobium in these alloys is quite apparent. It is the major eutectic phase-forming element, either NbC or $\delta/N_{3}N_{5}$, in two of the three alloys.

Structures of eutectic alloys are shown in Figure 24. The fibers (or in some cases lamellar plates) are obvious. Formed eutectically, they tend to be very stable and well-bonded to the matrix. The structure allows much of the potentially great strength of the hard fibers or plates to give great



Figure 23(a). IN-903A Showing Cubic γ ', Plate γ ", and Large Needles of δ -Ni₃Nb.



Figure 23(b). Same as 23a, Showing Another Cube Face of γ " and the δ Needles.

Figure 23. IN-903A with Over 5.5% Nb, Heat-Treated to Produce γ " From Bricknell, R., Personal Communication.

						w% \$	Solute				
Alloy	Morphology	ν _f	Ni	Co	Cr	AL	Nb	Mo	Ta	С	Other
Nitac	F	0.05	69		10	5			14.9	1.1	
Nitac 13	F		63	3.3	4.4	5.4			8.1	0.54	3.1 W,6.2Re,5.6V
CoTaC	F	0.10	10	65	10				14	Ι	
CoTaC 3 or 33*	F	0.10	10	56	20				13	1	
CoTaC-744	F		64	10	4	6	3.8	2		0.47	10W
γ/γ'- 6 (6%Cr)	L	0.3	71.5		6	2.5	20				
$\gamma/\gamma' - \delta$ (0%Cr)	L	0.3	76.5			2.5	21				
γ'/γ-Mo (AG-34)	F	0.26	62.5			6.3		31.2			
γ_δ	L	0.26	66.7				23.3				
r'-Ni-Ta	L	0.35	64.1			4.9		31			
Co,Cr-(Co,Cr)7C3	F	0.30		56.6	41					2.4	

Table IX.	Compositi	ons of So	me Directi	ionally Sc	lidified	Eutectic	Alloys.
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*1300 C, 2 h; 1000 C, 24 h, A.C.



Figure 24. Structure of a Fiber - Strengthened Eutectic Superalloy, and a Strength Comparison of Eutectic Alloys with Conventional Superalloys (32).

longitudinal creep capacity. Kahn (31) found the NbC fibers in COTAC-744 showed no visible damage following 2500 3-minute cycles between 1150 C and 2500 C. The lack of transverse grain boundaries reduces the potential of thermal fatigue crack initiation. The capability of some of these alloys to maintain load-carrying capability to high temperatures is shown in Figure 24. Further, they are vastly superior to conventional alloys in high-cycle fatigue (32).

Early work on carbide eutectics has showed partiality to TaC fiber structures. The potential now demonstrated by NbC shows two further advantages compared to tantalum, - greatly reduced density, and greatly reduced cost. Niobium is far more available and thus far less expensive than tantalum. However, a drawback appears to be limited oxidation and corrosion resistance, as discussed subsequently.

<u>Eutectic Alloy Summary</u>. Niobium is emerging as a key element in both fibrous and lamellar eutectic alloys. FCC NbC is the strengthening fibrous phase in a recent leading French alloy, CoTaC-744, and orthorhombic 6-Ni₃Nb is the strengthening lamellar phase in the $\gamma/\gamma' - \delta$ alloy system.

Effects of Niobium on Surface Stability

Discussion of this subject is divided into two sections, <u>oxidation</u> and <u>hot corrosion</u>. However, it is important to realize that hot corrosion can be considered simply as highly-enhanced oxidation caused by contaminant elements in the attacking atmosphere, such as sodium, potassium, vanadium and sulfur.

Oxidation

The broad, overall behavior of superalloys in oxidation has been noted in Figure 9. Their good high-temperature oxidation performance is a hallmark. It depends principally upon the ability to develop a stable, tenacious, tough, protective oxide film, usually provided by chromium (10-30%) and aluminum (1-6%).

Many of the multitude of elements added to superalloys affect oxidation resistance, improving or harming it for a wide variety of reasons. Small amounts of alloying elements can be significant. Thus, it is important to understand the effect of niobium, added in amounts from about 0.5 percent to 6 percent.

Unfortunately, there has been little clarifying study. Further, based on the properties of niobium as a pure metal, it is difficult to conjecture expected performance. Niobium itself readily forms a solid oxide in air, - usually a good sign for protective performance, - but the Nb_2o_5 formed (on unalloyed niobium) is of a fluffy, open-structured, nonprotective nature. Most superalloy metallurgists considered that niobium, while not observed to be harmful in oxidation, has also not been observed to be particularly helpful.

However, there have been two or three studies which, taken together, can shed some light. First, several years ago Davin, et al (34) studied Co-10Cr and Co-30Cr alloys to which a series of ternary elements, including 3 percent niobium were added. A portion of their results are shown in Figure 25. Inspection shows that molybdenum and niobium consistently promoted greatest attack. Further, the author singled out niobium as specifically causing catastrophic attack at high temperatures.

	W W N3 N3 17 5 T4 15 5 T4 15 15 15 15 15 15 15 15 15 15 15 15 15	Ac: 10	000 °(
	5 W Zr 0,5 Ce 1 Ar 0,1 B	10 Ni ¥ 11 Ti 1.5 Ta € 0.5 Y *	1100 1Nb) ℃ [
				*	3 Mo.
		1 Zr 10	5 ₩ _{Ni} * [1	200°C	70 mg/cm 1Nb
	05	Ce	1,5 ĭa		24 mg/cm
	1	A) 0,1 B			v
					0,5 Y ^r
0	1	2	3	4	5

WEIGHT GAIN, mg/cm² Figure 25. Air Oxidation of a Series of Co-30Cr-X Alloys for 5 Hours. After Davin (34),

In addition, Herchenroeder (35) studied an Fe-22Cr-20Ni-18Co-3.5Mo alloy (Haynes 556) as affected by a series of minor element additions. While it was found that some elements such as silicon and lanthanum can create a 5X improvement in resistance to dynamic oxidation testing conditions, niobium had a significantly deleterious effect. For instance, in one test (1093 C/2000 F, 100 hrs) zero Nb gave 51 mg/cm², 0.24 Nb 111 mg/cm², and 0.70 Nb 261 mg/cm², Niobium could overpower a beneficial effect found for tantalum but lanthanum could correct the problem. Also Seltran (36) states that "Columbium is clearly harmful in oxidation" and feels that poor performance of S-816 and WI-52 support the view.

In an interesting test by Meier (37) oxidation of a DS alloy strengthened by $\delta - \text{Ni}_3 \text{Nb}$ showed severe preferential attack of the $\text{Ni}_3 \text{Nb}$ at 700 C (1290 F) but not 1100 C (2000 F). Figure 26 illustrates this, and shows an $\text{Ni}_3 \text{Nb}$ denuded zone protects the alloy.

The message in this information appears clear. Niobium tends to harm oxidation resistance, based on laboratory tests, at or above about 2 percent. However, when attempts are made to correlate commercial alloy oxidation performance with niobium content, the results are not consistent. For instance IN-713C and MarM-432 (with 2% Nb) have excellent oxidation resistance, superior to well-known U-500 and many other alloys. On the other hand, U-700 (with no niobium) is excellent, and MarM-200 (1.8 Nb) is poor. Inspection of long lists of data for many alloys still generates no correlation to this writer. To cap the circumstance, IN-718 (with 5% Nb) has been found to have better oxidation resistance (as sheet at 1000 C/1800 F) than has Hastelloy X, an alloy utilized specifically for its fine oxidation performance. Undoubtedly, the amount and balance of other elements are generally of much more significance in causing oxidation resistance than niobium.



Figure 26. Oxidation of N1₃Nb - Strengthened DS Alloy at 1100 C (2000 F)³Above, and 700 C (1290 F), Below. Note Attack of N1₂Nb. After Meier (37).

Hot Corrosion

Hot corrosion attack is generally controlled in superalloys by keeping the Ti/Al ratio at 1:1 or better, maintaining chromium content at about 15 percent, restricting molybdenum to 1-2 percent, and by adding cobalt to nickel-based alloys. Again, evaluation of the effect of niobium on hot corrosion has been spotty, and it is difficult to be specific from further literature study or experience.

No work studying the effect of niobium in relatively simple laboratory systems (such as Ni/20Cr or Co-20Cr) has been seen by this author. Reviewing the behavior of complex commercial alloys with and without niobium also has been a relatively fruitless task.

For instance, Hosier and Harris (38) studied dry sulfidation (a mild corrosion-generating condition compared to condensing sodium sulfate) of a series of 26 alloys at 725 C (1340 F); four of the alloys (IN-625, IN-718, X-750 and 347 SS) contained niobium. Their resistance to corrosion was evenly scattered throughout the group. On the other hand, it has been noted that while 4 percent Nb-containing S-816 (at 870 C/1600 F) performs satisfactorily in hot corrosion, WI-52 (2% Nb) is poor at all temperatures (36). In a study for the US. Navy (39) niobium was present in the worst of twenty-four Ni-base alloys. Conversely we know that widely-used IN-738, universally well-regarded for its high corrosion resistance, contains almost 1 percent Nb. Further, an examination of the extensive information compiled by Stringer in AGARD document No. 200 (40) continues to frustrate; no trend can be seen.

<u>Surface Stability Summary</u>. Niobium can harm oxidation resistance significantly in simple laboratory alloys, but the effect can be masked in many commercial alloys. Whether niobium is helpful or harmful in hot corrosion performance is not known, but alloying effects from other elements appear to mask its behavior in any case.

Niobium in Superalloys; Summary of Behavior

Superalloys are complex FCC austenitic alloys strengthened by alloying additions participating in complex interreactions, - including substitutional solid solution, alloying effects at grain boundaries, formation of series of stable and reacting carbides, and formation of a series of coherent A_3 B-type compounds in nickel- and iron-base super-alloys. The alloying additions also affect and can control oxidation and hot corrosion properties significantly.

Niobium has been added in amounts up to about 6 percent to about half of the various complex cobalt, nickel and iron-nickel superalloys. Studies of a variety of superalloys show that niobium partitions to the austenitic alloy matrix, to the carbides, and, -when present, - the A_2B -type phases. Table

VIII summarizes the extensive phase partitioning possible by niobium, and Figure 27 identifies by microstructure the principal phase locations. From this review we find that niobium appears to have the following effect in strength and surface stability in these alloys:

- <u>Solid Solution Strengthening</u>. Due to limited solid solubility and unfavorable lattice mismatch, niobium has a small-to-negligible effect in simple solution strengthening of superalloys.
- <u>Carbide Participation</u>. Niobium forms highly stable MC carbides, which are significant in helping strengthen all types of superalloys, particularly cobalt base. Niobium forms NbC and also enters MC titanium carbides (viz., (Ti,Nb)C) in nickel alloys to affect/control their stability, and allow needed MC degeneration reactions to occur. However, the complex composition of these carbides prevents identifying whether niobium always acts as a stabilizer. In nickel alloys, 15-20 percent of the niobium present partitions to carbides.
- <u>Coherent Gamma Prime (Y') Phase.</u> In nickel alloys which form Y', niobium has been found to increase the volume fraction as much as 30 percent. Niobium in solution in both the gamma matrix and the Y' increases AFB energy, causing a significant increase in alloy yield strength, - vital in dislocation particle cutting by they' particles. Niobium also strengthens the Y' particles themselves.
- Coherent Gamma Double Prime (γ ") Phase. Niobium at levels up to about 6 percent is the principal strengthening element in Ni-Fe- ok Fe-Ni-base alloys. It forms γ ' and orthorhombic Ni₃Nb (γ "), the latter a unique and important strengthener in these superalloys. Upon over-aging, excessive Ni₃Nb δ -phase also can form, but it is not considered a strengthener.
- <u>Surface Stability</u>. Niobium appears to have a mixed or null effect on the oxidation and corrosion resistance of commercial resistance of commercial alloys and as tested in simple laboratory compositions. Some alloys which contain niobium demonstrate good ok excellent



Figure 27. The Role of Niobium (Columbium) in Conventional Superalloy Microstructure.

oxidation and hot corrosion behavior, although such behavior is usually ascribed to other alloying elements. Other alloys containing niobium can demonstrate very poor oxidation or hot corrosion resistance. In general there appears to be no evidence that niobium improves oxidation or corrosion resistance and simple laboratory alloy tests tend, more often than not, to show that niobium can be deleterious.

The Future in Superalloys

Perception

Over the years metallurgists have made small-to-medium additions of niobium to conventional superalloys; these additions have been significant, but have not been made on a systematic basis, or from broad and deep know-ledge about its behavior. Nonetheless, advancing technology, such as that evident in **DS** eutectics for very high temperature service, is generating an increased role in the superalloy sun for niobium.

The major reasons are the high-temperature stability of niobium metallurgical compounds, and its availability. FCC NbC and orthorhombic 6-Ni₃Nb form strengthening fibers or plates in eutectic alloys. In addition, the coherent complex hexagonal phase γ'' Ni₃Nb, which provides outstanding intermediatetemperature strengthening in iron-nickel alloys, appears prone for a greater role in high-performance alloys.

Action

These phenomena may be the top of a small iceberg, which should be exploited. For instance, it is perceived that:

- γ"-Νi₃Nb, singularly and modified by alloying, should be studied for application in a wider range of alloy base than up until now.
- Co₃Ti is moderately stable in cobalt austenitic alloy systems. Modifications with niobium might increase Co₃Ti stability.
- NbC will become a key fiber component in **DS** eutectics. Fundamental studies of its behavior, in both the "pure" and "alloyed" state are essential.
- δ-Ni₃Nb may have a future as a lamellar DS eutectic phase in austenitic superalloys. The fundamental properties of δ should be investigated thoroughly, and then studied for use in other alloy systems, and in combinations with NbC and γ".
- Oxidation and corrosion behavior will be harmed by niobium additions. The fundamentals of the effects of niobium on these properties must be established, so that clear definition of the role of niobium, both combined and uncombined, is obtained as a guide to the next item.
- Alloying of alloy systems and of the niobium compounds in them should be studied, with the objective of countering negative atmospheric-resistance effects,

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References

- 1. C. Sims, and W. Hagel (Eds.), The Superalloys, Wiley-Interscience, New York, 1972.
- 2. C. Sims, Proceedings MICON '78 Conference, Houston, Texas, April, 1978.
- 3. R. Guard and J. Westbrook, Trans AIME, 215, 1959, p. 807.
- E. Gua and F. Mai, "The Strengthening Effect of Niobium in Ni-Cr-Ti Type Wrought Superalloy," in <u>Proc. 4th Int'l Conference in Superalloys</u>, pp. 431-8, ASM Press, Metals Park, Ohio, 1980.
- 5. C. Sims, <u>J. Metals</u>, 3, October, 1966, p. 1119.
- 6. J. Mihalisin, et al., Trans AIME, 242, 1968, pp. 2399-2414.
- 7. R. Decker and C. Bieber, pp. 120-28 in <u>ASIM STP No. 262</u>, ASIM, Philadelphia, PA., 1959.
- 8. H Beattie and W. Hagel, Trans. AIME, 221, February 1967, pp. 28-35.
- 9. G. Harris and N. Child, p. 67, in <u>High-Temperature Steels and Alloys for</u> <u>Gas Turbines</u>, Iron & Steel Inst, London, July 1952.
- 10. Private Communication, Carl Lund, 1981.
- 11. E Ault and M. Donachie, Effects of 1600F Exposure on the Microstructure of MarM/Alloy 421, Martin Metals Co., 1969.
- 12. The Superalloys, p. 1, Wiley-Interscience, New York, 1972.
- 13. O. Kriege and J. Baris, Trans ASM, 62, 1969, p. 195.
- 14. P. Thornton, et. al., <u>Met Trans</u>, 1, 1970, pp. 207-218.
- 15. J. Drapier, J. de Brouwer and D. Coutsouradis, <u>Cobalt</u>, 27, June, 1965, pp. 59-72.
- 16. J. Drapier, A. Davin and D. Coutsouradis, Cobalt, 37, 1975, pp. 32-42.
- 17. G. Bouse and P. Schilke, Proc. 4th Int'l Conf. on Superalloys, ASM Press, Metals Park, Ohio, 1980.
- N Stoloff, The Superalloys, pp. 79-112, Wiley-Interscience, New York, 1972.

1219

- D. Muzyka, The Superalloys, pp. 113-143, Wiley-Interscience, New York, 1972.
- 20. D. Paulonis, et. al., Trans ASM. 62, 1969, p. 611.
- 21. R. Cozar and A. Pineau, Met Trans, 4, 1973, p. 47.
- 22. W. Quist, et. al., Met Trans., 2, 1971, pp. 825-832.
- 23. U. L. Eiselstein, p. 62 in, <u>ASTM STP No. 369</u>, ASTM, Philadelphia, PA, 1965.
- 24. W. Boesch and H. Canada, J. of Metals, 21, 1969, p. 34.
- 25. D. Keiser and H. Brown, in ERDA Report ANCR-1292 UC-25, February, 1976.
- H. L. Eiselstein, "Age-Hardenable Nickel Alloys," U.S. Patent 3,046,108, July 24, 1962.
- 27. F. Rizzo and J. Buzzanell, J. of Metals, 21, 1969, p. 24.
- 28. J. Stroup and R. Heacox, J. of Metals, 21, 1969, p. 46.
- 29. Private Communication, V. Biss, AMAX, 1981.
- 30. In <u>Proceedings: First COST-50 Conference</u>, Brussels, September, 1979.
- T. Kahn, et. al., "Cotac 744; An Optimized DS Composite for Turbine Blades," pp. 531-540 in Proc. 4th Int'l Conference on Superalloys, ASM Press, Metals Park, OH, 1980.
- N Stoloff and D. Duquette, p. 788 in, <u>ASTM Spec. Tech. Publ. 675</u>, ASTM, Philadelphia, PA, 1979.
- 33. R. Bricknell and D. Woodford, Met. Trans., in process.
- 34. A. Davin, D. Coutsouradia and L. Habraken, <u>Cobalt</u>, 29, June, 1967, p. 35.
- 35. R Herchenroeder, in Int'l Conf. on Behavior of High-Temperature Alloys in Aggressive Environments, Petten, NL, October, 1979.
- 36. A. M. Beltran, Cobalt, 46 1970, p. 3-14.
- 37. Private Communication, J. Mier, Univ, of Pittsburgh, 1981.
- Hozier and Harris, Paper No. 167 in <u>NACE Int'l Corrosion Forum</u>, Chicago, Iil, November 3-7, 1980.
- C. Sims, P. Bergman and A. Beltran, <u>Naval Engineers J.</u>, April, 1970, pp. 39-52.
- J. Stringer, in <u>AGAARD-AG-200</u>, NATO, 7 rue Ancelle, 92220 Neuilly sur Seine, France, August, 1975.